





## CH-53K FREQUENTLY ASKED QUESTIONS

| Specification Weight Empty  | 43,878 lb                         | 19,000 kg   | Engines - Three (3) General   |
|---|-----------------------------------|-------------|---|
| Basic Design GW (BDGW)  | 60,000 lb                         | 27,200 kg   | Class Turboshaft Engines  |
| Structural Design GW (SDGW)   | 60,000 lb                         | 27,200 kg   | Installed Takeoff Rating (Sea Level Standard)   |
| Alternate Design Gross Weight (ADGW)<br>for Maximum Internal Load   | 74,000 lb                         | 33,600 kg   | Installed Takeoff Rating<br>(3000 ft 91.5°F or 914m 33°                               |
| Maximum Design Gross Weight (MDGW)<br>for Maximum External Load   | 88,000 lb                         | 39,900 kg   | Installed Intermediate Ratin<br>(Sea Level Standard)                                  |
| How fast does the aircraft fly?   |                                   |             | Installed Intermediate Ratin  |
| Velocity horizontal   | 170 kta                           | s 315 kph   | (3000 ft 91.5°F or 914m 33°   |
| Velocity dive   | 196 kta                           | s 363 kph   | Installed Maximum Continuou   |
| Velocity @ Maximum Continuous Power & A<br>(Sea Level Standard)   | DGW 158 kta                       | s 293 kph   | (Sea Level Standard)  Installed Maximum Continuou                                     |
| /elocity @ Maximum Continuous Power & A<br>(Sea Level 103°F or Sea Level 39.4°C)  | DGW 164 kta                       | s 304 kph   | (3000 ft 91.5°F or 914m 33°<br>1 transmission limited                                 |
| /elocity @ Maximum Continuous Power & A<br>(3000 ft 91.5°F or 914m 33°C)  | DGW 161 kta                       | s 298 kph   | How many rotor blades d dimensions? What are th                                       |
| Velocity @ Maximum Continuous Power & M<br>(Sea Level Standard)   | IDGW 133 kta                      | s 246 kph   | Blades (Main Rotor)<br>Blades (Tail Rotor)  |
| Velocity @ Maximum Continuous Power & M<br>(Sea Level 103°F or Sea Level 39.4°C)  | IDGW 135 kta                      | s 250 kph   | Geometric Chord (Main Roto<br>Geometric Chord (Tail Rotor                             |
| Velocity @ Maximum Continuous Power & M<br>(3000 ft 91.5°F or 914m 33°C)  | IDGW 126 kta                      | s 233 kph   | Diameter (Main Rotor)<br>Diameter (Tail Rotor)  |
| Miles Carlo de la companya de la constanta de |                                   |             | Normal Tip Speed (Main Rot  |
| What is the hover & service ceiling ca  |                                   | 20,000 14   | Normal Tip Speed (Tail Roto   |
| Hover Out of Ground Effect Gross Weight (Sea Level Standard)  | 88,000 lb                         | 39,900 kg   | Normal Operating Rotor Revo   |
| Hover Out of Ground Effect Gross Weight<br>(Sea Level 103°F or Sea Level 39.4°C)  | 88,000 lb                         | 39,900 kg   | What is the capability of the Hook Rating - Center (Single                            |
| Hover Out of Ground Effect Gross Weight<br>(3000 ft 91.5°F or 914m 33°C)  | 85,100 lb                         | 38,600 kg   | Hook Rating - Center (Single<br>Hook Rating - Fwd/Aft<br>Dual Point (30/70 & 70/30) w |
| Hover In Ground Effect Gross Weight<br>(Sea Level Standard)   | 88,000 lb                         | 39,900 kg   | Dual Point (30/70 & 70/30) w,<br>Dual Point (40/60 & 60/40)                           |
| Hover In Ground Effect Gross Weight<br>(Sea Level 103°F or Sea Level 39.4°C)  | 88,000 lb                         | 39,900 kg   | What is the capability of t   |
| Hover In Ground Effect Gross Weight<br>(3000 ft 91.5°F or 914m 33°C)  | 88,000 lb                         | 39,900 kg   | Floor Loading 30<br>Standard USMC 40"x 48" 2  |
| Service Ceiling @ ADGW (International Standard Atmosphere)  | 16,000 ft                         | 4,880 m     | Wooden Pallets  |
|   | 13,200 ft                         | 4,020 m     | Full 463L Pallets 10  |
|   | 4°CJ                              |             | Half 463L Pallets 5,  |
| Service Ceiling Id ADGW<br>(International Standard Atmosphere + 2   |                                   |             | Tactical Bulk Fuel 3 Delivery System  |
| (International Standard Atmosphere + 2  | pability of the                   | e aircraft? |   |
| (International Standard Atmosphere + 2<br>What is the mission performance cap   | 27,000 lb                         | 12,200 kg   |   |
| (International Standard Atmosphere + 2<br>What is the mission performance cap<br>External Lift Mission Payload,<br>@ 110 nm (204 km) Radius of Action   |                                   |             | What type of armor syste  Armor Panels Provides  Cockpit Non-rem                      |
| What is the mission performance cap<br>External Lift Mission Payload,<br>@ 110 nm (204 km) Radius of Action<br>Internal Lift Mission Payload,   | 27,000 lb<br>13,100 lb<br>>400 nm | 12,200 kg   | What type of armor syste  Armor Panels Provides                                       |

| What powerplant does the CH-53K employ?                                 |                         |                        |
|---|-------------------------|------------------------|
| Engines - Three (3) General Electric T408-G<br>Class Turboshaft Engines | E-400 7,500 F           | HP (5,593 kw)          |
| Installed Takeoff Rating<br>(Sea Level Standard)                        | 17,700 shp¹             | 13,200 kW¹             |
| Installed Takeoff Rating<br>(3000 ft 91.5°F or 914m 33°C)               | 16,200 shp              | 12,100 kW              |
| Installed Intermediate Rating<br>(Sea Level Standard)                   | 16,000 shp <sup>1</sup> | 12,200 kW <sup>1</sup> |
| Installed Intermediate Rating<br>(3000 ft 91.5°F or 914m 33°C)          | 12,462 shp              | 9,300 kW               |
| Installed Maximum Continuous Power Rating<br>(Sea Level Standard)       | 13,000 shp <sup>1</sup> | 11,900 kW¹             |
| Installed Maximum Continuous Power Rating (3000 ft 91.5°F or 914m 33°C) | 12,100 shp              | 9,000 kW               |

How many rotor blades does the aircraft have? What are the dimensions? What are the operating speeds and RPMs?

| Blades (Main Rotor)                      | 7                |               |
|--|------------------|---------------|
| Blades (Tail Rotor)                      | 4                |               |
| Geometric Chord (Main Rotor)             | 2.72 ft          | 0.829 m       |
| Geometric Chord (Tail Rotor)             | 1.48 ft          | 0.451 m       |
| Diameter (Main Rotor)                    | 79 ft            | 24.1 m        |
| Diameter (Tail Rotor)                    | 20 ft            | 6.10 m        |
| Normal Tip Speed (Main Rotor)            | 762 ft/sec       | 232 m/sec     |
| Normal Tip Speed (Tail Rotor)            | 754 ft/sec       | 230 m/sec     |
| Normal Operating Rotor Revolutions Per M | linute (Main Ro  | otor) 184 RPM |
| Normal Operating Rotor Revolutions Per M | linuta (Tail Rot | or) 720 RPM   |

What is the capability of the External Cargo System?

| Hook Rating - Center (Single Point)            | 36,000 lb | 16,300 kg  |
|--|-----------|------------|
| Hook Rating - Fwd/Aft                          | 25,200 lb | 11,400 kg  |
| Dual Point (30/70 & 70/30) w/o fuel management | 13,500 lb | 6,100 kg   |
| Dual Point (30/70 & 70/30) w/ fuel management  | 27,000 lb | 12,200 kg  |
| Dual Point (40/60 & 60/40)                     | 36,000 lb | 16,300 kg. |

What is the capability of the Internal Cargo System?

| Floor Loading                            | 300 lb/ft²           | 1,470 kg/m²          |
|--|----------------------|----------------------|
| Standard USMC 40"x 48"<br>Wooden Pallets | 2,500 (x qty 6) lb   | 1,100 (x qty 6) kg   |
| Full 463L Pallets                        | 10,000 (x qty 2) lb  | 4,500 (x qty 2) kg.  |
| Half 463L Pallets                        | 5,000 (x qty 5) lb   | 2,300 (x qty 5) kg.  |
| Tactical Bulk Fuel<br>Delivery System    | 3 x 800 gallon tanks | 3 x 3030 liter tanks |

#### What type of armor system does the CH-53K have?

| Armor Panels             | Provides protection to small arms fire              |  |
|--------------------------|---|--|
| Cockpit                  | Non-removable pilot and copilot seat & wing armor   |  |
| Cabin                    | Non-structural panels within the floor ramp & walls |  |
| Mission Configurable Yes |   |  |

| What type of Flight Control System does the CH-53K have?<br>What are the primary hardware elements and the basic system capabilities? |  |  |
|---|--|--|
| Flight Control System   | Triplex redundant fly by wire system. The primary control system consists of 3 flight control computers (FCCs), 3 inertial measurement units (IMUs), and 3 air data computers (ADCs) |  |
| Pilot Controls  | Active unique trim sidearm inceptor<br>Active displacement trim collective inceptor<br>Passive unique trim pedals  |  |

#### What flight director modes does the CH-53K have?

Flight director modes Indicated Airspeed (IAS) Hold

Groundspeed Hold

Pitch Barometric Altitude Hold

Vertical Speed Hold Altitude Pre Select **Heading Select** Departure/Wave-Off

Flight Management System Long Range Navigation

Approach to Point

#### What are the primary flight control modes in the CH-53K?

Primary flight Primary Flight Control System -

Control modes Provides rate command, attitude hold in pitch/roll

Automatic Flight Control System -

Provides attitude command, velocity hold in pitch/roll

#### What defensive Electronic Counter Measures (DECM) does the CH-53K have?

State-of-the-Art Aircraft Survivability suite

#### Who are the suppliers for each of the major systems and subsystems?

Airframe, Cockpit/Cabin Spirit Airframe, Aft Transition GKN

Airframe, Sponson, Tail Rotor Pylon

Airframe, Main Rotor Pylon Landing Gear

Environmental Control System Flight Control Servos

Engines

Avionics Management System Integrated Fuel Systems / Hydraulics

Engine Air Particle Protection System Fuel Cell

Integrated Vehicle Health

Management System Internal Cargo

Tail Drive System

Main Rotor & Tail Rotor Servos Electrical Power Systems

Main Gearbox Housing

Cockpit Seats

Active Inceptor System Ramp and Door Actuation Main Rotor Blade Spar Assembly

Data Concentrator Unit / Blade Fold Distributor Curtiss Wright

Troop Seats Inlet Ducts

Main Blade Spar/Tail Rotor Blade/Main Rotor

Blade Skin to Core/Tail Rotor Blade Flex Beam Hexcel Air Data Computer

Environmental Control System Ducts Hydraulic Blade Fold System / Main Rotor

Damper / Rotor Brake Hydraulic Module / Main Gear Box & Nose Gear Box Oil Cooler

Albany International

Aurora Heroux-Devtek

Collins Aerospace Collins Aerospace General Electric Aviation

Collins Aerospace Eaton Aerospace Donaldson Amfuel

Collins Aerospace

DRS

Collins Aerospace Collins Aerospace

STADCO (machinings)

Safran Labinal Power Systems Wellman (castings) /

BAE

BAE Systems

Arkwin

Cobham Composite Products

East West

Meggitt Polymers and

Composites

Penny & Giles

Royal Engineered Composites

Triumph

#### What are the Reliability and Maintainability metrics for the CH-53K?

| Mission Reliability  | 89%            |
|--|----------------|
| Sortie Generation Rate   | 2.6            |
| Mean Flight Hours Between Operational Mission Failure<br>Design Controllable (MFHBOMFDC) | s -<br>31.5 hr |
| Mean Flight Hours Between Failures -<br>Design Controllable (MFHBFDC)                    | 1.7 hr         |
| Maintenance Man-Hour per Flight Hours -<br>Organizational (MMH/FHORG)                    | 11.7 hr        |
| Mean Time To Repair (MTTR)   | 1.5 hr         |
| Mean Corrective Maintenance Time for Operational Mission Failures (MCMTOMF)              | 1.9 hr         |

#### What is the logistics footprint?

Weight Status < 84,000 lb < 38,100 kg Volume Status < 12,000 ft<sup>3</sup> < 340 m<sup>3</sup>

#### What are the capabilities of the Diagnostics/Integrated Vehicle Health Management System (IVHMS)?

Fault Detection 95% Fault Isolation 90%

(w/o interconnects and Government Furnished Equipment)

Centralizes the monitoring and reporting of faults detected by onboard sensors and

embedded electronics

Performs bearing monitoring, and rotor track

and balance

Generates data to support ground-based

automated logistics

Incorporates two Integrated Vehicle Health Management Units (IVHMU) and Sikorsky Ground Based Application (SGBA)



#### How big is the CH-53K aircraft (blades and tail unfolded)?

| Length | 99 ft   | 30.2 m |
|--------|---------|--------|
| Width  | 17.5 ft | 5.3 m  |
| Height | 28.3 ft | 8.6 m  |

#### How big is the CH-53K cabin?

| Length | 30 ft  | 9.1 m |  |
|--------|--------|-------|--|
| Width  | 8.6 ft | 2.6 m |  |
| Height | 6.5 ft | 2.0 m |  |

#### How many people can the CH-53K carry?

34 Crashworthy Seats Pilot, Copilot, 2 Crew Chiefs, 30 Marines

#### What is the material breakdown for the CH-53K aircraft?

| Composite | 17.7% |
|-----------|-------|
| Aluminum  | 20.9% |
| Steel     | 15.5% |
| Titanium  | 15.9% |
| Other     | 30.0% |

#### What type of armament system does the CH-53K have?

Machine Guns Three (3) 50-caliber GAU-21 machine guns in the

door window, forward cabin window, and ramp

#### What type of crash capability does the CH-53K have?

| Vertical     | 20g |
|--------------|-----|
| Longitudinal | 20g |
| Lateral      | 10g |

Seats Crash attenuating air crew & troop seats

#### What type of environmental control system is on the aircraft?

Cockpit and Avionics/Electrical Bays Cooled Cockpit and Cabin Heated

## What are the operational design environment limits for the CH-53K aircraft?

| Temperature | -40 to +122 °F | -40 to +50 °C |
|-------------|----------------|---------------|
| Altitude    | 17,000 ft      | 5,180 m       |

## How long does it take to fold the main rotor blades and the tail rotor pylon?

2 minutes

## What ship platforms does the CH-53K have the ability to support?

Fully compatible on LHD, LHA-6 class ships

# What fixed wing aircraft can the CH-53K aircraft be air transported on? How long does it take to breakdown and restore the aircraft?

| C-5  | 2 x CH-53K |
|------|------------|
| C-17 | 1 x CH-53K |

Disassembly time Elapsed Time 13.5 hrs, Maintenance Man Hours 70.5 hrs
Reassembly time Elapsed Time 22.5 hrs, Maintenance Man Hours 130 hrs

## What type of gearbox does the CH-53K have and what are the ratings at Normal Operating Rotor speed (184 RPM)?

| Gearbox type                       | Split Torque                  |                              |
|------------------------------------|-------------------------------|------------------------------|
| Takeoff Rating                     | 17,700 shp<br>(136.2% Torque) | 13,200 kw<br>(136.2% Torque) |
| Intermediate Rating                | 16,000 shp<br>(123.1% Torque) | 11,930 kw<br>(123.1% Torque) |
| Maximum Continuous<br>Power Rating | 13,000 shp<br>(136.2% Torque) | 9,700 kw<br>(136.2% Torque)  |

## How much fuel does the CH-53K carry? How many tanks? How long does it take to refuel the aircraft?

| Internal fuel capacity        | 2,286 gallons<br>(@ 6.8lb/gallon =<br>15,545 lb)   | 8,653 liters<br>(@0.82 kg/liter =<br>7,095 kg) |
|-------------------------------|--|--|
|                               | 2 cells, per sponson   |  |
| Internal<br>aux fuel capacity | 2,400 gallons<br>(16,320 lb)   | 9,085 liters<br>(7,450 kg)                     |
|                               | 3 x 800 gallon tanks   | 3 x 3,028 liter tanks                          |
| Refueling                     | Aerial or ground refueling is expected to take up to 18.5 minutes (depending on starting fuel level in each fuel cell and the associated environmental conditions) |  |

## Describe the CH-53K Displays and Controls, Communication and Navigation systems.

Displays & Controls Avianies management system

| Displays & Controls | (5 portrait MFDs, 2 CDUs, 2 IPCs, and 2 DTUs) Dual Data Concentrator Units   |
|---------------------|--|
| Communications      | 3 AN/ARC-210 RT-1851A [C] Multi Function<br>Radios with SATCOM capability<br>AB/APX-123 Identification Friend or Foe (IFF)<br>AN/USQ-140(V)1[C] MIDS LVT<br>Digital Secure Wireless ICS<br>4 Wired and 4 Secure Wireless stations  |
| Navigation          | Dual LN-251 Embedded GPS Inertial Navigation<br>Systems (EGI)<br>Advanced Digital Antenna Production (ADAP) GPS<br>Antenna System<br>3 Air Data Computers<br>Dual RT-1805/APN LPIA Radar Altimeters<br>NAV-4500 VHF Omni-Directional range/instrument<br>landing system/marker beacon (VOR/ILS/MB)<br>AN/AQ-29A Forward Looking Infrared Radar (FLIR)<br>PMA-209 TAWS - Embedded Terrain |

#### What comprises the Electrical Power System on the CH-53K?

Awareness Warning System (eTAWS)

Electrical Power System 2 Main Generators 75KVA each

Auxiliary Power Unit Generator 45KVA 2 Main 350 amp AC/DC converters Backup 200 amp converter

24 AH battery

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6900 Main Street, Stratford, Connecticut 06615 USA

+1 (800) WINGED-S (946-4337)

+1 (203) 386-3029 (International) www.lockheedmartin.com/sikorsky

